DEPARTMENT OF CIVIL ENGINEERING, IIT MADRAS

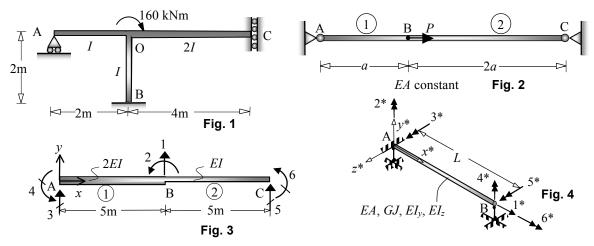
CE 3310: ADVANCED STRUCTURAL ANALYSIS

Class: 3/4 B Tech (Civil)END SEMESTER EXAMNovember 28, 2011Maximum Marks: 50'F' SlotTime Limit: 180 minutes

Answer Question 1 (10 marks) and any FIVE other questions (8 marks each).

1. Answer **any four** of the following $(2\frac{1}{2} \text{ marks each})$:

(a) Distribute the applied moment in the frame shown in Fig. 1 appropriately to the various frame elements and hence draw the bending moment diagram. Also sketch the deflected shape.



(b) Consider the axially loaded system in Fig. 2. Derive an expression for the deflection at B, in terms of P, a and EA and draw the axial force diagram.

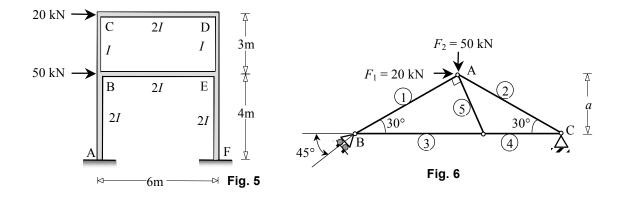
(c) Fig. 3 depicts the global coordinates in a non-prismatic fixed beam ABC (fixed at ends A and C). Using the conventional stiffness approach, generate the 4×4 transformation matrices, T^1 and T^2 , for the two elements, marking the linking global coordinates, and hence, the fixed $[F_{f_A}]$

end force vector, $\mathbf{F}_{\mathbf{f}} = \begin{bmatrix} \mathbf{F}_{\mathbf{f}\mathbf{A}} \\ \mathbf{F}_{\mathbf{f}\mathbf{R}} \end{bmatrix}$, due to a uniformly distributed load of 12 kN/m on the beam.

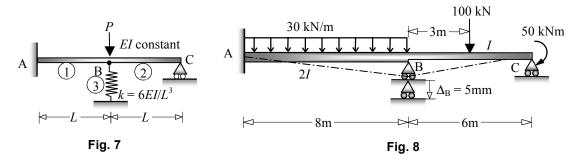
(d) Consider an element AB in a plane truss, connecting the joint A(0,1) to joint B(3,5), where the global axes (x,y) dimensions are expressed in metres. Find the axial deformation in the bar, given that A is connected to a rigid support and B deflects by 0.5mm and -1.5mm in the x and y directions respectively. Assume an element axial rigidity of 50000 kN.

(e) Generate the reduced element stiffness matrix $\tilde{\mathbf{k}}_*$ of a typical prismatic space frame element with 6 degrees of freedom, with local coordinates and properties, as depicted in Fig. 4 [No need for any derivation]. Also, generate the element fixed end force vector \mathbf{F}_{*f} given that the element is subject to a uniformly distributed load of q_0 per unit length.

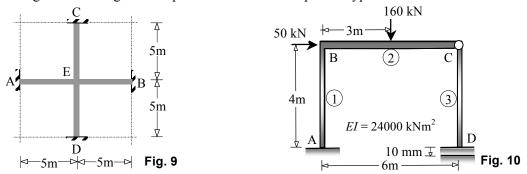
- 2. Analyse the laterally loaded single bay two-storeyed frame shown in Fig. 5 by an approximate force method (assuming the beams to be infinitely rigid) and draw the bending moment diagram. Hence, applying the unit load method, find an approximate value of the horizontal drift in the frame at the top, assuming flexural rigidity $EI = 10000 \text{ kNm}^2$.
- 3. Analyse the laterally loaded single bay two-storeyed frame in Fig. 5 by the slope-deflection or moment distribution method, taking advantage of anti-symmetry in the response. Draw the bending moment diagram. Also sketch the deflection profile.



- 4. (a) Generate the force transformation (T_F) matrix (of size 5×2), and hence find the bar forces in the five-bar truss, loaded as shown in Fig. 6.
 (b) In an unloaded condition, it is given that the five bars in Fig. 6 undergo thermal elongations given by e₀ = [2, 2, 3, 1, 1]^T mm. Find the deflections D₁ and D₂ at A.
- 5. Analyse the spring-supported beam shown in Fig.7 by any method of your choice (stiffness or flexibility) and hence (i) find the deflection at B and (ii) draw the bending moment diagram of the beam. The spring stiffness is given by $6EI/L^3$, where EI is the flexural rigidity of the beam.



- 6. Analyse the two-span continuous beam shown in Fig.8 by any <u>matrix</u> method of your choice (stiffness or flexibility) and draw the bending moment diagram. Assume $EI = 27000 \text{ kNm}^2$.
- 7. Consider the symmetric grid system (plan view) shown in Fig. 9, with all the elements subjected to uniformly distributed gravity loading of 12 kN/m and having uniform flexural rigidity EI = 27000 kNm² and torsional rigidity GJ = 0.2EI. Taking advantage of symmetry and adopting any method of your choice, find the deflection at the centre E and draw the bending moment diagram and probable deflected shape of a typical element AE.



8. Analyse the portal frame shown in Fig. 10 (with an internal hinge at C, loading and support settlement of 10mm at D, as shown) by the reduced stiffness method, ignoring axial deformations, to find (a) the horizontal deflection at B, and (b) the hogging moment at B. Assume $EI = 24000 \text{ kNm}^2$ for all members.